Annexure III

<u>Technical Specification for :</u> Coaxial rotorcraft with flight controller, telemetry and GNSS based navigation

Specification		Complied/ Not Complied	Reference Page No.
Configuration	Coaxial rotor		
Control mechanism	Variable Rotor collective and cyclic pitch for both rotors		
Max. Rotor diameter	1000 mm		
Rotor material	Honeycomb and carbon fiber composites		
Airframe	Composites		
Max. dimensions with blades removed	Cylindrical volume; D=600mm, H=250mm		
мтоw	8 kg		
Max. Payload	2.5 kg (Removable design)		
Empty weight (w/o battery)	< 3 kg		
Drive system	Geared or direct drive with BLDC motors		
Power	Li-ion 12S Battery pack (Removable)		
Max operable altitude	3000m above MSL		
Endurance at 3000m	15 minutes		
Programmable flight controller development board	 Attitude, position and heading control Customizable collective/cyclic mixing (CCPM) In flight tuning capability with remote gain adjustment 		
GNSS receiver	GPS, GLONASS, NavIC + RTK support		
Guidance Tools	Programmable planner with waypoint navigation		
Ground station	2.4GHz Radio transceiver with telemetry		
Radio control & telemetry range (LOS)	1000m		
Gust resistance	Class 8 (18m/s)		

Terms and conditions		Reference Page No.
1. Controller board architecture and telemetry/guidance interface to be submitted		
2. Payload tray and battery should be removable easily		
3. Lead time allowable for delivery is strictly 4 weeks		

(Note: It is mandatory for the bidders to provide the compliance statement in tabular column format along with catalogue page number (comply/not comply) for the Above points with document proof as required. Failing which bidders will be technically disqualified)